

Tutorial for Fiber Reinforced Piping (FRP) Modeling and Analysis using CAEPIPE

The following are the Steps for FRP Modeling and Analysis using CAEPIPE.

General

FRP piping has gained wide acceptance in many industries due to its lightweight nature, superior corrosion resistance, temperature capabilities and mechanical strength. Several manufacturers produce different types of FRP pipes and fittings and provide technical assistance to their customers on design matters through installation.

FRP piping can be modeled in CAEPIPE. CAEPIPE will then calculate deflections, element forces & moments, support loads and stresses.

Tutorial

Snap shot shown below is a sample model for FRP Modeling and Analysis

The image shows two windows from the CAEPIPE software. The left window is the 'Layout' window, displaying a table of piping components. The right window is the 'Graphics' window, showing a 3D isometric view of the piping system with various fittings, valves, and supports.

#	Node	Type	DX (ft/in)	DY (ft/in)	DZ (ft/in)	Matl	Sect	Load	Data
1	Title = Tutorial for FRP Piping								
2	10	From	337'8"	210'5-3/4"	-264'6"				Anchor
3	10	Location							Flange
4	20				-3'0-1/8"	FW	24	L1	Anchor
5	30				-2'3-5/8"	FW	24	L1	Flange
6	32	Valve			-0'8-1/4"	CS	24	L1	Flange
7	35				-2'9-1/2"	FW	24	L1	
8	35	Location							
9	38				-1'1-1/4"	FW	24	L1	Flange
10	38	Location							Conc mass
11	39				-0'6-1/4"	FW	24	L1	
12	40	Bend			-2'0"	H12	24F	L1	
13	40B	Location							
14	40B	Location							
15	50		2'9"			FW	24	L1	
16	60		0'6"			FW	24	L1	
17	70		0'6"			FW	24	L1	Guide
18	70	Location							X restraint
19	80		1'0"			FW	24	L1	
20	90		1'0-1/8"			FW	24	L1	Flange
21	100	Valve	0'8-1/4"			CS	24	L1	Flange
22	100	Location							
23	110		1'3-1/8"			FW	24	L1	Reinf tee
24	120		1'5-1/2"			FW	24	L1	
25	130		0'5"			FW	24	L1	
26	140		0'5"			FW	24	L1	
27	150		0'10"			FW	24	L1	
28	160		0'5"			FW	24	L1	Reinf tee
29	170		0'9"			FW	24	L1	
30	175		0'5"			FW	24	L1	Rest Supp
31	175	Location							Restraint

Step 1:

First define FRP materials required for piping system through Layout window > Misc > Materials. In the Material List window shown on the screen, double click on an empty row to input a new material or on a material description to edit the material properties.

Step 2:

In the Material dialog shown, enter the FRP material properties as given below.

Material # 1 [X]

Material name

Description

Type

Density (lb/in³)

Nu

Joint factor

Yield strength (psi)

Tensile strength (psi)

The material name can be up to five alpha-numeric characters. Enter description, density and Poisson's ratio. You need to select "FR: Fiber Reinf. Plastic (FRP)" from the Type drop-down combo box before you click on the Properties button.

Step 3:

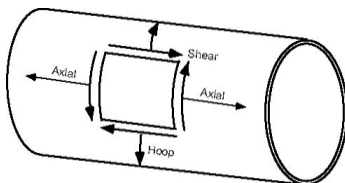
Click on the Properties button, you are shown the table below where you can enter temperature-dependent properties. Additionally, you can also define the Hoop, Torsional and Axial allowable stresses so that CAEPIPE can use them to compare with calculated stresses under the FRP "Sorted Stresses" results.

Material Properties [X]

#	Temp (F)	Axial Mod. (psi)	Hoop Mod. (psi)	Shear Mod. (psi)	Alpha (in/in/F)	Hoop All. (psi)	Torsional All. (psi)	Axial All. (psi)
1	70	1.30E+6	2.00E+6	1.20E+6	16.30E-6	3191	1450	3191
2	125	1.30E+6	2.00E+6	1.20E+6	16.30E-6	3191	1450	3191
3	392	1.30E+6	2.00E+6	1.20E+6	5.74E-6	3191	1450	3191
4								

FRP Material Moduli

CAEPIPE requires three moduli for the FRP material:



- Axial or Longitudinal (this is the most important one)
- Hoop (used in Bourdon effect calculations). If this modulus is not available, use axial modulus.

- Shear or Torsional. If this modulus is not available, use engineering judgment in specifying 1/2 of axial modulus or a similar value. Note that a high modulus will result in high stresses, and a low modulus will result in high deflections.

For FRP bends, a Flexibility factor of 1.0 is used unless you override it by specifying a Flexibility factor inside the bend dialog. Also, for FRP bends, CAEPIPE uses a default SIF of 2.3. You can override this too by specifying User-SIFs at the bend end nodes (A and B nodes).

Step 4:

After defining the FRP material properties, Section Properties and Loads required for the stress analysis, complete the stress layout. Save the model and Analyze through Layout window > File > Analyze.

Step 5:

Upon successful analysis, CAEPIPE will show the calculated stresses, deflections, support loads, element forces and support load summary. Each item can be seen under the respective title in Results. FRP element stresses can be seen, sorted or unsorted.

#	Hoop				Max Long				Min Long				Torsion			
	Node	Stress (psi)	Allow (psi)	Stress/Allow	Node	Stress (psi)	Allow (psi)	Stress/Allow	Node	Stress (psi)	Allow (psi)	Stress/Allow	Node	Stress (psi)	Allow (psi)	Stress/Allow
1	20	3537	3191	1.11	110	12925	3191	4.05	110	-10994	3191	3.45	1110	1041	1450	0.72
2	20	3537	3191	1.11	1160B	3832	1450	2.64	1160B	-2968	1450	2.05	1100	1041	1450	0.72
3	30	3537	3191	1.11	40A	3498	1450	2.41	40A	-2658	1450	1.83	1140	696	1450	0.48
4	1560	3537	3191	1.11	1160A	2940	1450	2.03	110	-4964	3191	1.56	1120	696	1450	0.48
5	1210	3537	3191	1.11	160	5930	3191	1.86	160	-4553	3191	1.43	1130	696	1450	0.48
6	1280	3537	3191	1.11	38	5557	3191	1.74	120	-4372	3191	1.37	1145	696	1450	0.48
7	300	3537	3191	1.11	38	5557	3191	1.74	1160A	-1899	1450	1.31	1150B	666	1450	0.46
8	250	3537	3191	1.11	230	5391	3191	1.69	160	-3842	3191	1.20	1155	666	1450	0.46
9	32	3537	3191	1.11	1170	5218	3191	1.64	100	-3782	3191	1.19	1110	508	1450	0.35
10	1235	3537	3191	1.11	1150A	2221	1450	1.53	90	-3711	3191	1.16	2010	332	1450	0.23
11	35	3537	3191	1.11	210	4792	3191	1.50	38	-3653	3191	1.14	2000	332	1450	0.23
12	290	3537	3191	1.11	1790A	2134	1450	1.47	38	-3653	3191	1.14	2120	323	1450	0.22

#	Node	Hoop (psi)	Axial (psi)	Bending (psi)	Longitudinal Max (psi)	Longitudinal Min (psi)	Torsional (psi)
1	10	3537	-1408	33	-1375	-1441	0
	20	3537	-1408	21	-1387	-1429	0
2	20	3537	952	2696	3648	-1744	80
	30	3537	952	1297	2249	-345	80
3	32	3537	952	412	1364	540	80
	35	3537	952	1992	2944	-1040	80
4	35	3537	952	1992	2944	-1040	80
	38	3537	952	4605	5557	-3653	80
5	38	3537	952	4605	5557	-3653	80
	39	3537	952	3296	4248	-2344	80
6	40A	1504	420	3078	3498	-2658	32
	40B	1504	588	1531	2119	-943	15
7	40B	3537	1366	1639	3006	-273	36
	50	3537	1366	448	1814	918	36
8	50	3537	1366	448	1814	918	36
	60	3537	1366	476	1842	900	36

#	Node	Displacements (global)					
		X (inch)	Y (inch)	Z (inch)	XX (deg)	YY (deg)	ZZ (deg)
1	10	0.000	0.000	0.000	0.0000	0.0000	0.0000
2	20	0.000	0.000	0.000	0.0000	0.0000	0.0000
3	30	-0.067	-0.011	-0.050	-0.0230	0.1723	0.0086
4	32	-0.092	-0.014	-0.068	-0.0231	0.1725	0.0087
5	35	-0.198	-0.026	-0.129	-0.0114	0.0682	0.0191
6	38	-0.211	-0.028	-0.153	-0.0054	-0.0476	0.0232
7	39	-0.206	-0.028	-0.164	-0.0030	-0.1169	0.0251
8	40A	-0.206	-0.028	-0.164	-0.0030	-0.1170	0.0251
9	40B	-0.045	-0.013	-0.134	-0.0052	-0.2620	0.0321
10	50	-0.026	-0.007	-0.078	-0.0065	-0.2952	0.0285
11	60	-0.013	-0.003	-0.038	-0.0073	-0.2953	0.0239
12	70	0.000	0.000	0.000	-0.0082	-0.2778	0.0174
13	80	-0.001	0.001	0.053	-0.0099	-0.2238	0.0055

#	Node	Tag	FX (lb)	FY (lb)	FZ (lb)	MX (ft-lb)	MY (ft-lb)	MZ (ft-lb)
1	10	1	-426	62939	-205	0	0	
2	20		-7969	-2406	-46765	-4152	26357	1580
3	590		-19	791	-1973	-431	-4	-261
4	890		-14	234	309	-205	-6	-65
5	990		-9	214	404	-197	-5	-3
6	1090		-4	295	64	-229	-2	14
7	1100		-1323	577	-1034	-259	-1005	-4503
8	1410	6		-1094	-447	-1184	0	-18
9	1510		-3	-1083	-438	-1162	2	12
10	1710		-2	-1067	-419	-1093	3	12
11	1910	2		-1061	-416	-1088	-4	-13
12	2000		235	1235	-508	-315	337	1436

#	Node	Axial (lb)	y Shear (lb)	z Shear (lb)	Torque (ft-lb)	Inplane(ft-lb)		Outplane(ft-lb)	
						Moment	SIF	Moment	SIF
1	10	-62939	-409	1	0	-205	1.60	0	1.60
	20	-62939	409	1	0	-206		3	
2	20	-16174	-1997	-7967	-1580	-4358		26360	
	30	-16174	-1371	-7967	-1580	-482	1.60	8018	1.60
3	32	-16174	-760	-7967	-1580	251	1.60	2541	1.60
	35	-16174	-1	-7967	-1580	1313		-19702	
4	35	-16174	-1	-7967	-1580	1313		-19702	
	38	-16174	299	-7967	-1580	1148	1.60	-28499	1.60
5	38	-16174	354	-7967	-1580	1148	1.60	-28499	1.60
	39	-16174	495	-7967	-1580	927		-32649	
6	40A	-16174	-7967	-495	-1580	32649	2.30	927	2.30
	40B	-7967	16174	-1403	-723	16236	2.30	-566	2.30
7	40B	-7967	1403	16174	-723	-566		-16236	
	50	-7967	1607	16174	-723	-1695		-4106	
8	50	-7967	1607	16174	-723	-1695		-4106	

Load combination	FX (lb)	FY (lb)	FZ (lb)	MX (ft-lb)	MY (ft-lb)	MZ (ft-lb)
Sustained	0	-426	0	-205	0	0
Operating1	1	-426	62939	-205	0	0
Sustained+Seismic 1	36	-408	36	-197	17	0
Sustained-Seismic 1	-36	-444	-36	-214	-17	0
Operating1+Seismic 1	37	-408	62974	-197	17	0
Operating1-Seismic 1	-35	-444	62903	-214	-18	0
Maximum	37	-408	62974	-197	17	0
Minimum	-36	-444	-36	-214	-18	0
Allowables	0	0	0	0	0	0

Stiffness matrix formulated internally in CAEPIPE and the formulas used for computing different stresses are given below for quick reference.

Stiffness matrix

The stiffness matrix for a pipe is calculated using the following terms:

Axial term = L / EA

Shear term = shape factor $\times L / GA$

Bending term = L / EI

Torsion term = $L / 2GI$

where L = length, A = area, I = moment of inertia, E = elastic modulus, G = shear modulus

For an isotropic material, $G = E / 2(1 + \nu)$, where ν = Poisson's ratio,

For a FRP material, E = axial modulus and G is independently specified (i.e., it is not calculated using E and ν).

The hoop modulus and FRP Poisson's ratio are only used in Bourdon effect calculation where,

Poisson's ratio used = FRP Poisson's ratio input \times (axial modulus / hoop modulus)

FRP Stresses

$$\text{Hoop stress: } S_H = \frac{PD}{2t_m}$$

$$\text{Axial stress: } S_A = \frac{PD}{4t_m} + \frac{F}{A}$$

$$\text{Bending stress: } S_B = \frac{\sqrt{(i_i M_i)^2 + (i_o M_o)^2}}{Z}$$

$$\text{Torsional stress: } S_T = \frac{M_t}{2Z}$$

$$\text{Longitudinal maximum} = \text{Axial} + \text{Bending} = S_A + S_B$$

$$\text{Longitudinal Minimum} = \text{Axial} - \text{Bending} = S_A - S_B$$

where

P = pressure

D = outside diameter

t_m = minimum thickness

= nominal thickness x (1 - mill tolerance/100) - corrosion allowance

i_i = in-plane stress intensification factor

i_o = out-of-plane stress intensification factor

M_i = in-plane bending moment

M_o = out-of-plane bending moment

M_t = torque

Z = section modulus

F = axial force

A = cross-section area